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Laboratory



Center for Nondestructive
Evaluation



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Testing Technologies, Inc.*



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Detecting Cracks under Bushings in Aircraft Structures Using FG RFEC & SSEC Technique

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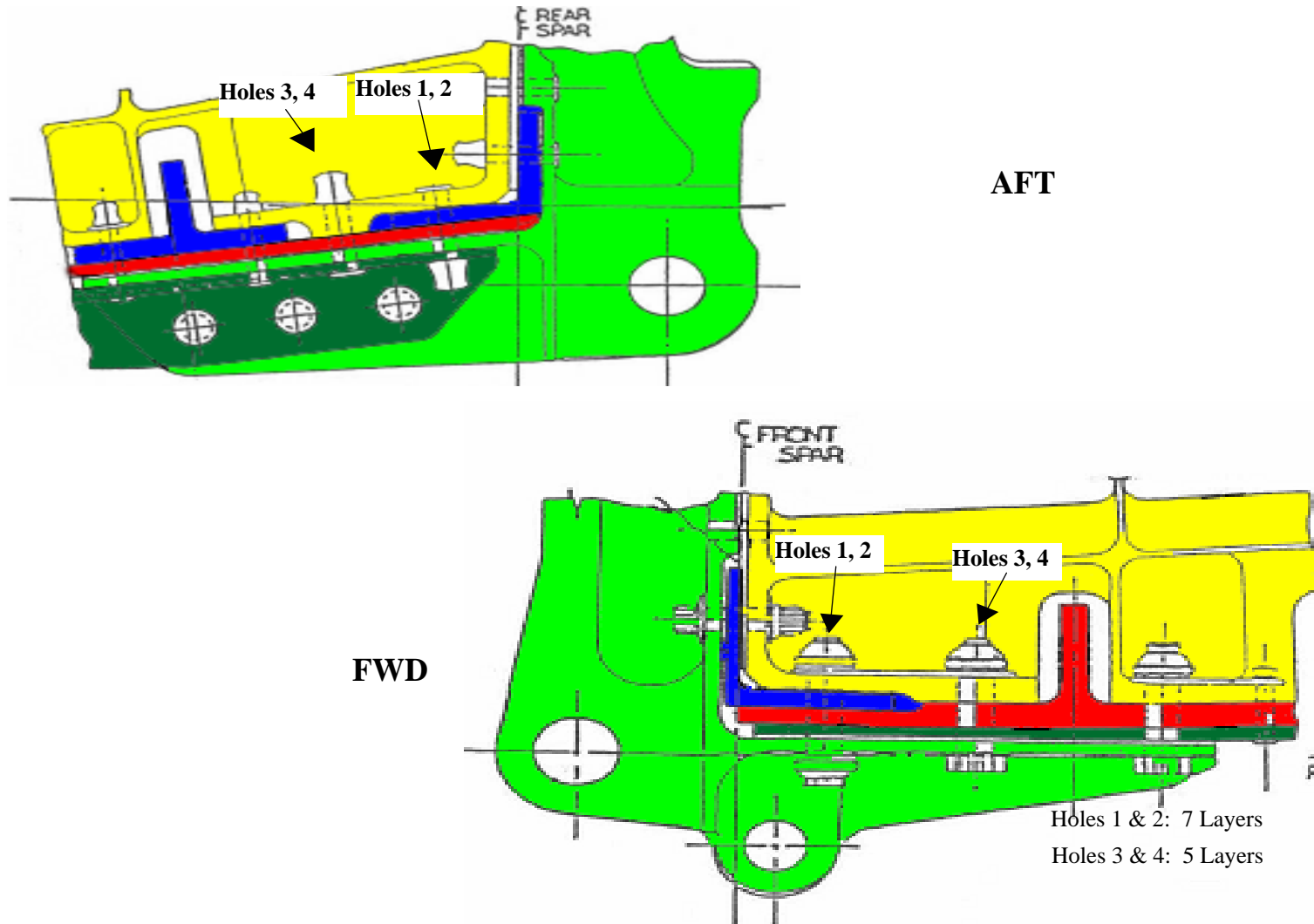


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- 1. A challenge to detect cracks in cross-bolt holes when the cracks are located beneath repair bushings.**
- 2. Removal of a bushing is costly and may bring new damage to the holes.**
- 3. There has been a demand for techniques that is capable of detecting cracks in cross-bolt holes through a bushing.**

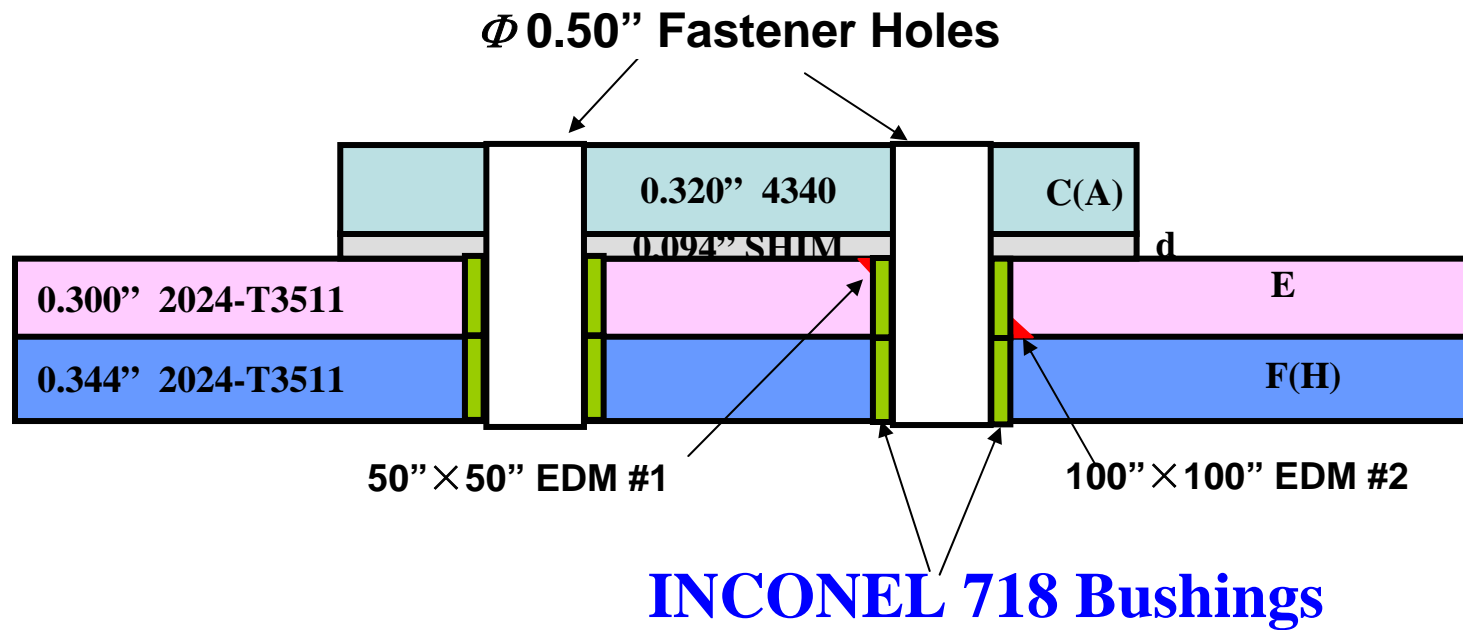


Target Aircraft Component: Wing Attach Fitting Inspection through bushing.





Test Standard





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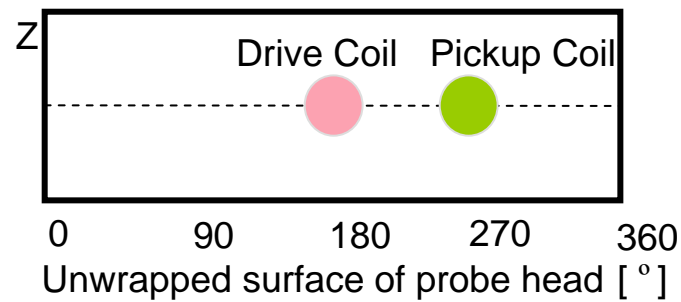
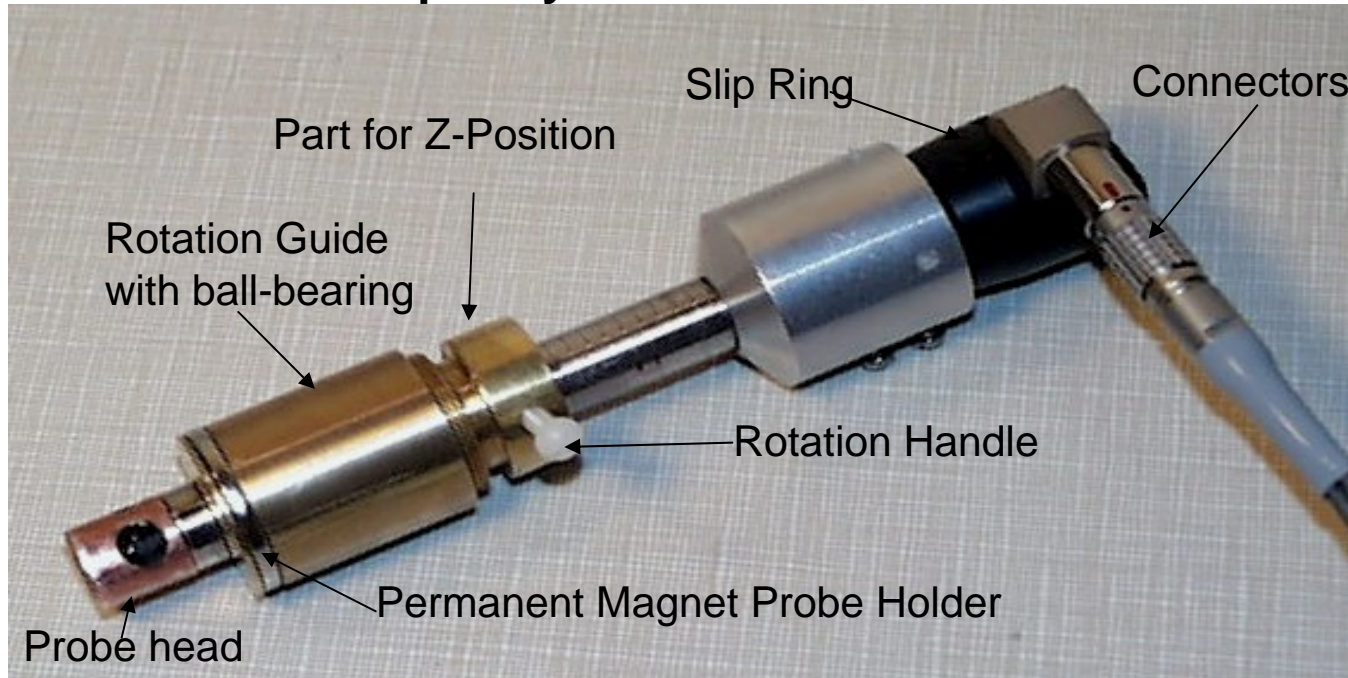
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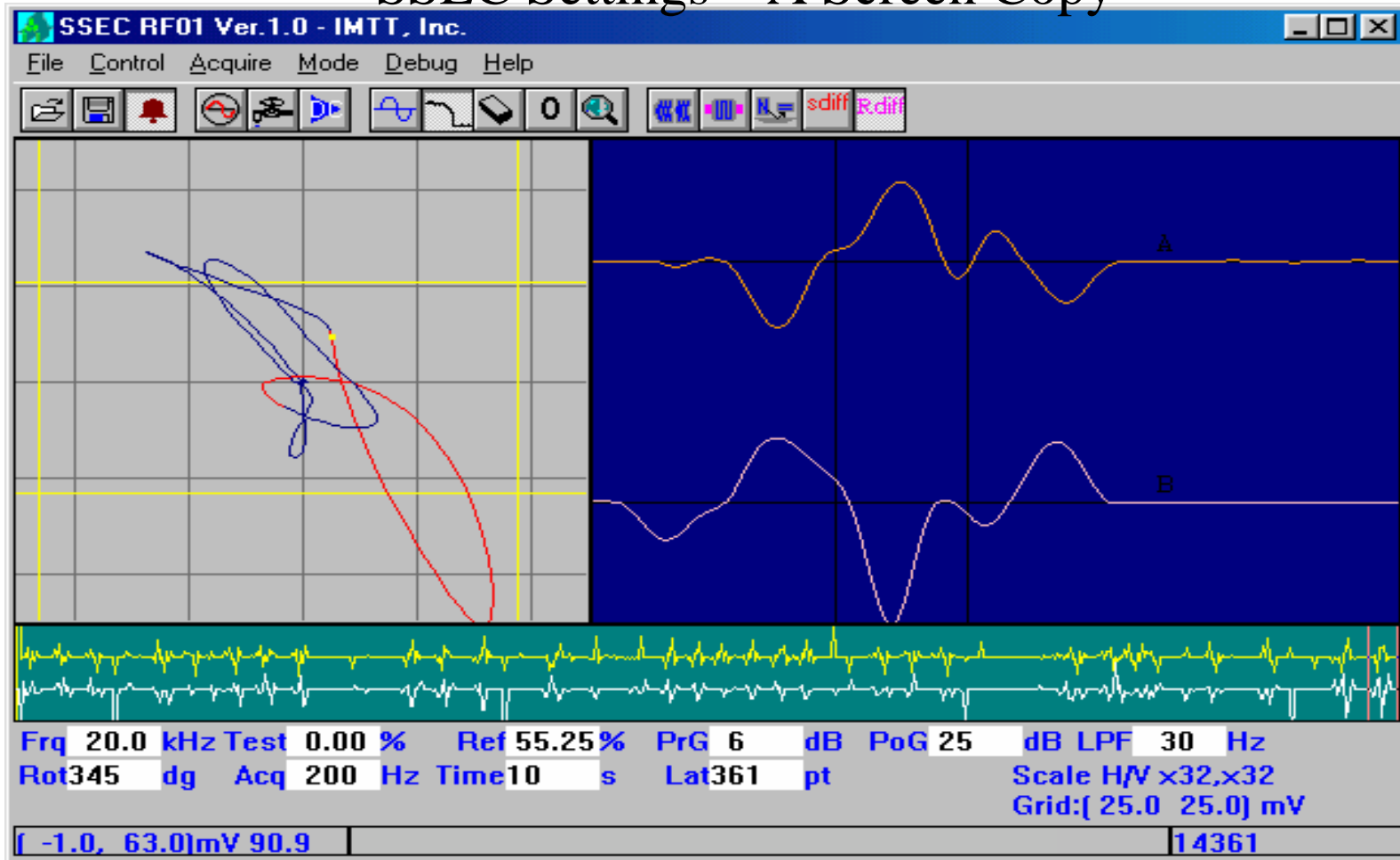
Probe Developed:
Test Frequency:

RF BSH0.5
10 kHz - 80 kHz





SSEC Settings – A Screen Copy



Simulated Reflection Differential Mode with step = 5 points
Digital LPF, cutoff frequency = 1.0 Hz



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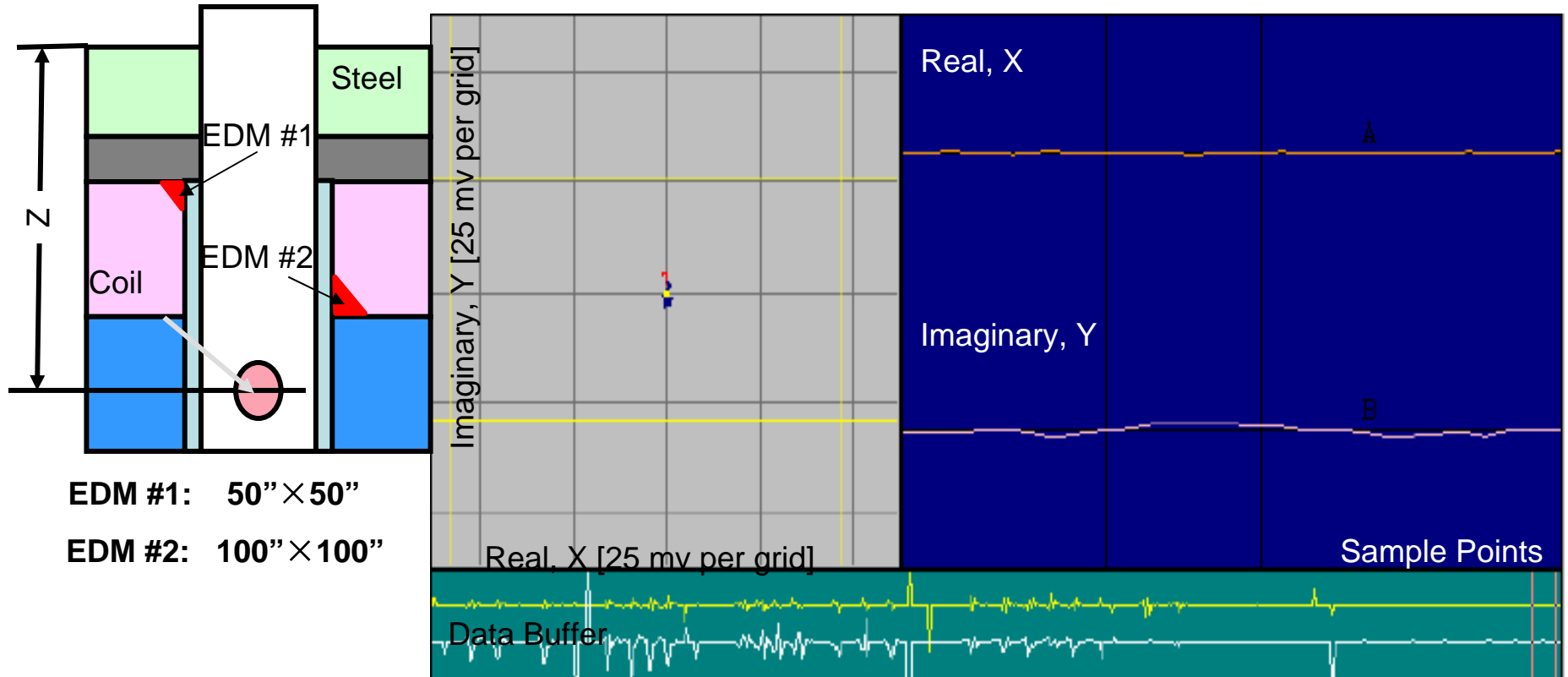


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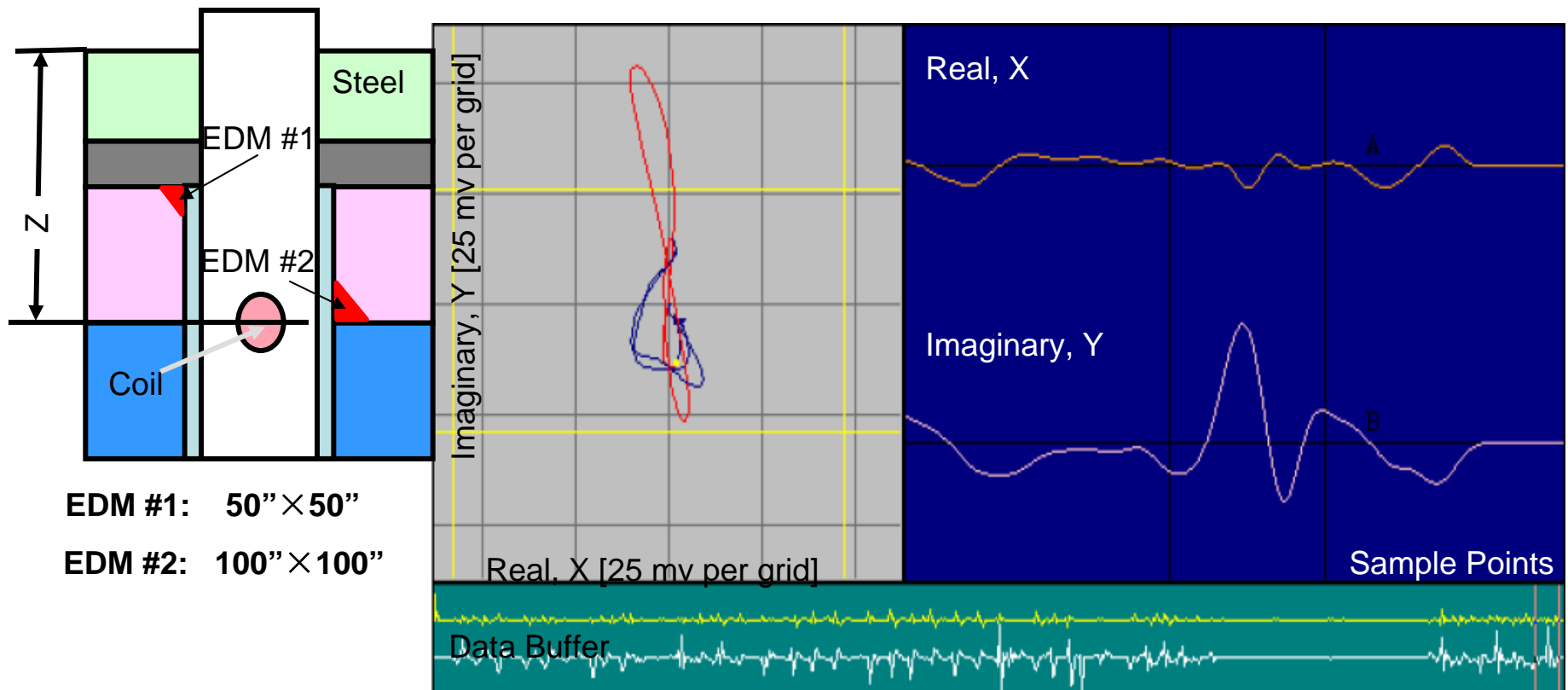
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Probe Position 1 $Z = 0.885''$

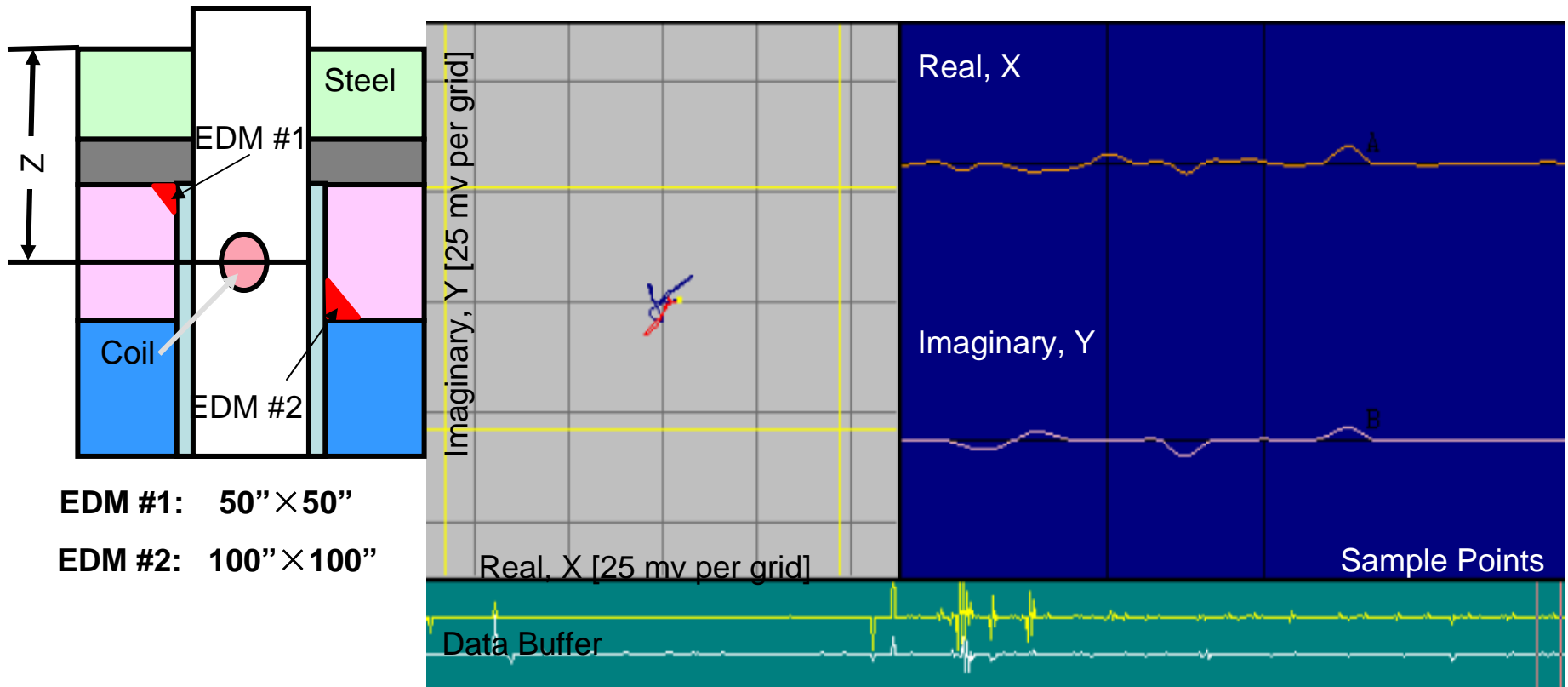


Position 2 $Z = 0.715''$

EDM #2 can be detected from $Z=0.685''$ to $Z=0.760''$

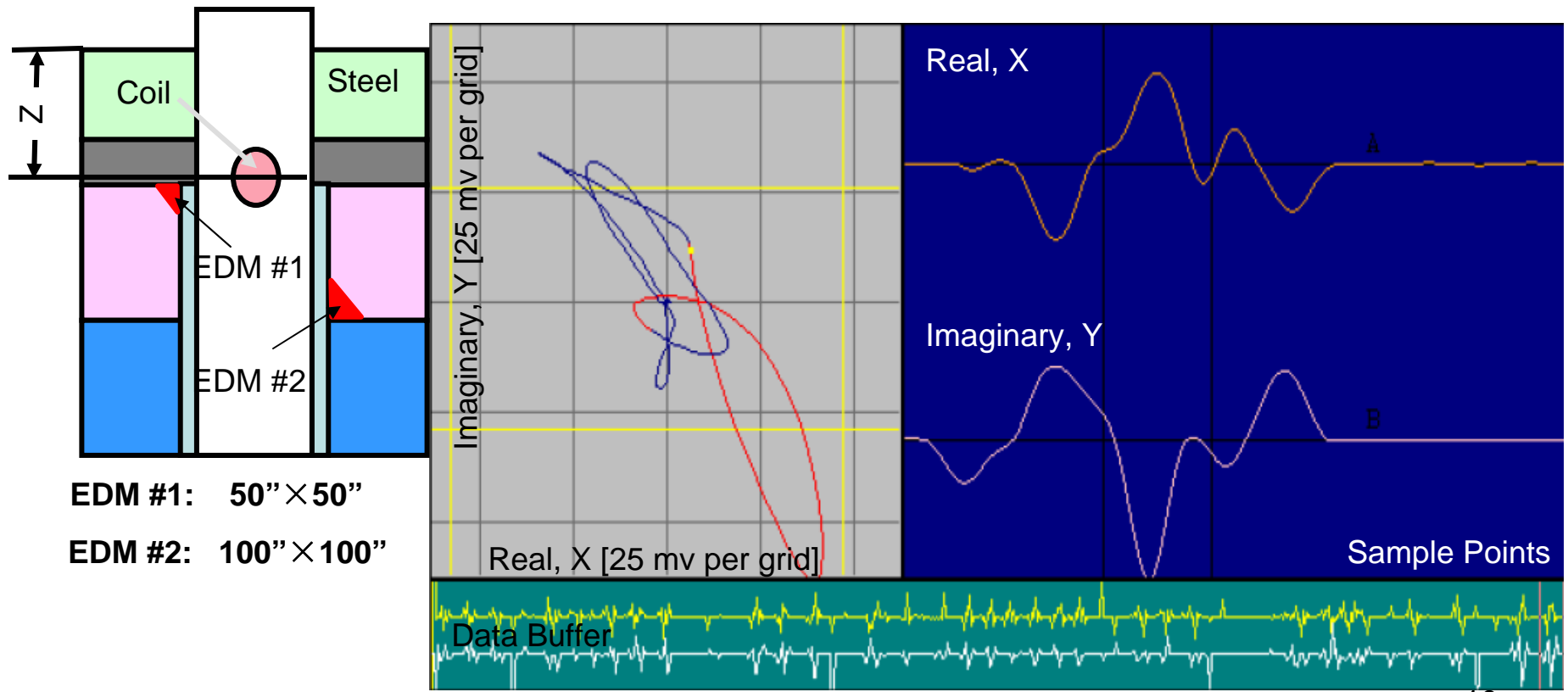


Probe Position 1 $Z = 0.565''$



Position 2 $Z = 0.395''$

EDM #1 can be detected from $Z=0.395''$ to $Z=0.415''$





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Conclusions

- 1. The two EDM Notches made on layer E are detected with high S/N ratios.**
- 2. The permanent magnet probe holder and the ball-bearing rotation guide help with noise reduction and ease of operation.**
- 3. Comparing the signal responses of the two notches the 0.100" × 0.100" EDM Notch, Notch #2, can be detected in a large range of probe position Z, see Page 8.**
- 4. The 4340 steel layer enhances the signal response from the 0.050 " × 0.050" EDM notch, Notch #1. However, it moves the location for optimal detection towards steel side, see Page 10.**



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Future Work

- 1. Motorize probe rotation**
- 2. Automatic C-Scan and Real-Time Imaging**
- 3. Consider thickness variation of bushing**
- 4. Try detection of bore-hole crack with and without an installed bushing.**
- 5. Commercialization**
- 6. Field testing and POD study**